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# Strength of Materials 

For Second Stage Students<br>In Mechanic \& Mechatronics Dept.

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## Classification of Engineering Mechanics



## Strength of materials

It deals with the relations between externally applied forces or loads and the internal effects in the body.

When materials are loaded, they first deform before actual failure takes place. Hence before selecting any material for engineering purpose, it is important to know the behavior of the material under the action of loads.

Unit: - it is defined as the numerical standard used to measure the qualitative dimension of a physical quantity.

TABLE 1-1 Basic quantities in the SI metric unit system.

| Quantity | SI unit | Other metric units |
| :--- | :--- | :--- |
| Length | Meter $(\mathrm{m})$ | Millimeter $(\mathrm{mm})$ |
| Time | Second $(\mathrm{s})$ | Minute $(\mathrm{min})$, hour $(\mathrm{h})$ |
| Force | Newton $(\mathrm{N})$ | $\mathrm{kg} \cdot \mathrm{m} / \mathrm{s}^{2}$ |
| Mass | Kilogram $(\mathrm{kg})$ | $\mathrm{N} \cdot \mathrm{s}^{2} / \mathrm{m}$ |
| Temperature | Kelvin $(\mathrm{K})$ | Degrees Celsius $\left({ }^{\circ} \mathrm{C}\right)$ |
| Angle | Radian $(\mathrm{rad})$ | Degree $\left({ }^{\circ}\right)$ |

TABLE 1-2 Basic quantities in the U.S. Customary unit system.

| Quantity | U.S. Customary unit | Other U.S. units |
| :--- | :--- | :--- |
| Length | Foot $(\mathrm{ft})$ | Inch (in.) |
| Time | Second $(\mathrm{s})$ | Minute $(\mathrm{min})$, hour (h) |
| Force | Pound (lb) | $\mathrm{kip}^{\mathrm{a}}$ |
| Mass | Slug | $\mathrm{lb}^{\mathrm{s}} \mathrm{s}^{2} / \mathrm{ft}$ |
| Temperature | Degrees Fahrenheit $\left({ }^{\circ} \mathrm{F}\right)$ |  |
| Angle | Degree $\left({ }^{\circ}\right)$ | Radian (rad) |

TABLE 1-3 Prefixes for SI units.

| Prefix | SI symbol | Factor |
| :--- | :---: | :--- |
| Giga | G | $10^{9}=1000000000$ |
| Mega | M | $10^{6}=1000000$ |
| Kilo | k | $10^{3}=1000$ |
| Milli | m | $10^{-3}=0.001$ |
| Micro | $\mu$ | $10^{-6}=0.000001$ |



## Customary Units of Length

1 foot (ft) $=12$ inches (in)
1 yard (yd) $=3$ feet (ft)
1 yard (yd) $=36$ inches (in)
1 mile (mi) = 1,760 yards (yd)
1 mile $(\mathrm{mi})=5,280$ feet $(\mathrm{ft})$

## Converting VOLUME Units



VOLUME conversions use powers of 3 , and usually create very large results. $3 \mathrm{~m}^{3}=? \mathrm{~cm}^{3}$ Need to $\times 100^{3} \quad 3 \times 100 \times 100 \times 100=3000000 \mathrm{~cm}^{3} \sqrt{ }$

## Mechanical properties of material

The following are considered as the most important properties of engineering materials

1) Elasticity
2) Plasticity
3) Ductility
4) Malleability
5) Brittleness
6) Toughness
7) Hardness

Any material cannot possess all the above properties because the different properties oppose each other. Hence the engineering Materials can be classified as follows depending upon their Mechanical properties
1)Elastic Materials: - These are materials which undergo deformation due to application of forces and once the forces are removed the material regains its original shape.
2)Plastic materials: - These are materials which do not regain their original shape even after the external loads acting on the material are removed.
3)Ductility: - It is a measure of a material's ability to undergo significant plastic deformation before rupture, which may be expressed as percent elongation or percent area reduction from a tensile test.
4)Malleable materials: - These are Materials which can be extended in two directions easily or in simple terms, materials which can be beaten into thin sheets.
5)Brittle materials: - these are materials which do not undergo any deformation before failure when external forces act on them.
6)Tough materials: - These are materials which can resist sudden loads or shock loads without showing any fracture on failure
7)Hard material: - These are materials that have the ability to resist surface abrasion or indentation.

Various tests are carried out on engineering materials to assess their mechanical properties in a material testing laboratory. They are

1) Tension test
2) Compression test
3) Impact test
4) Shear test
5) Torsion test
6) Bending test
7) Fatigue test
8) Hardness test

## Stress

Whenever some external forces act on a body it sets up a deformation and the body offer some resistance against deformation. this resistance per unit area to deformation is known as "stress".

## Types of Stresses

The various types of stresses may be classified as:

## 1- Simple or Direct Stresses

i. Tension Stress ii. Compression Stress iii Shear Stress

## 2- Indirect Stresses

i. Bending Stress ii. Torsion Stress

3- Combined Stresses, any possible combination 1 and 2.

## Simple stresses

Simple stress is offen called direct stress because it developse under direct loading conditions. That is, simple tension and simple compression occur when the applied force or load, is in line with the axis of the member (fig. 1.1 and 1.2) and simple shear occurs, when equal, parallel and opposite forces tend to cause a surface to slide relative to the adjacent surface (fig. 1.3).


Fig. 1.1. Tensile stress


Fig. 1.2. Compressive stress


Fig. 1.3. (a) Rivet resisting shear.
(b) Rivet failure due to shear.

When simple stress (tension and compression) ( $\sigma$ ) (sigma) develops, we can calculate the magnitude of the tensile or compression stress by,

$$
\text { stress }=\frac{\text { force }}{\text { area }}=\sigma_{\mathrm{t}, \mathrm{c}}=\frac{\mathrm{p}}{\mathrm{~A}}
$$

Where $\quad \begin{aligned} \sigma & =\text { Stress (also called intensity of Stress) } \mathrm{kN} / \mathrm{m}^{2} \text { or } \mathrm{N} / \mathrm{mm}^{2} \\ \mathrm{P} & =\text { External Force or Load }(\mathrm{kN} \text { or } \mathrm{N}) \\ \mathrm{A} & =\text { Cross-Sectional Area of the body }\left(\mathrm{m}^{2} \text { or } \mathrm{mm}^{2}\right)\end{aligned}$

It may be noted that in cases of either simple tension or simple compression, the areas which resist the load are perpendicular to the direction of forces. When a member is subjected to simple shear, the resisting area is parallel to the direction of the force. Common situations causing shear stresses are shown in fig. 1.3 and 1.4. we can calculate the magnitude of the shear stress by,

$$
\tau=\frac{\mathrm{P}}{\mathrm{~A}}
$$



Fig. 1.4.(a) Punch approaching plate;
(b) Punch shearing plate;
(c) Slug showing sheared area.

## Stress Units

Psi (Pound per square inch)
Kpsi (Kilo pound per square inch)

## In SI Unit

$\mathrm{Pa}\left(\right.$ Pascal $\left.=\mathrm{N} / \mathrm{m}^{2}\right)$
$\mathrm{Kpa}($ Kilo $\mathrm{Pa}=1000 \mathrm{~Pa})$
$\operatorname{Mpa}\left(\operatorname{MegaPa}=10^{6}\right)$

## Strain

Any element in a material subjected to stress is said to be strained. The strain $(\varepsilon)$ is the deformation produced by stress. The various types of strains are explained below:

1. Tensile strain
2. Compressive strain
3. Volumetric strain
4. Shear strain

## 1- Tensile Strain

A piece of material, with uniform cross-section, subjected to a uniform axial tensile stress, will increase its length from $(\mathrm{L})$ to $(\mathrm{L}+\delta \mathrm{L})$, and the increment of length $\delta \mathrm{L}$ is the actual deformation of the material. The fractional deformation or the tensile strain is given by


## 2- Compressive Strain

Under compressive forces, a similar piece of material would be reduced in length from $L$ to $(\mathrm{L}-\delta \mathrm{L})$. the fractional deformation again gives the strain $\mathcal{E}_{\mathrm{c}}$.

$$
\varepsilon_{\mathrm{c}}=\frac{\delta \mathrm{L}}{L}
$$


(Original Length)

## 3－Shear Strain

In case of a shearing load，a shear strain will be produced which is measured by the angle through which the body distorts．


Note that shear stress is tangential to the are over which it acts．As the bottom face of the block is fixed，the face ABCD will be distorted to $\mathrm{ABC}_{1} \mathrm{D}_{1}$ through an angle $(\phi)$ as a result of force P as shown in figure above．And shear strain $(\phi)$ is given by：

$$
\text { Shear Strain }=\phi=\frac{\mathrm{DD} 1}{\mathrm{AD}}=\frac{\mathrm{TQ}}{h}
$$

## 4－Volumetric Strain

It is defined as the ratio between change in volume and original volume of the body．

Volumetric Strain $\varepsilon_{\mathrm{v}}=\begin{aligned} & \delta \mathrm{v} \\ & -\quad \text { Cha回 } g \text { 回 } i \text { 回回回回 }\end{aligned}$

## Elastic Limit

We have already discussed that whenever some external system of force acts on a body, it undergoes deformation. If the external forces, causing deformation, are removed the body springs back to its original position.

Beyond the elastic limit, the material gets into plastic stage and in this stage the deformation does not entirely disappear, on the removal of the force. But as a result of this, there is a residual deformation even after the removal of the force.


## Hook's Law

Hook's law states that when a material is loaded within elastic limit, the stress is proportional to the strain produced by the stress. This means the ratio of the stress to the corresponding strain is a constant within the elastic limit. This constant is known as Young's Modulus or Modulus of Elasticity and is denoted by (E). But the ratio of shear stress to the corresponding shear strain within the elastic limit, is known as Modulus of Rigidity or Shear Modulus. This is denoted by (G).
$\mathrm{E}=\underline{\text { TensileStress }}$ or $=\underline{\text { Compressive Strass }}$ or $=\frac{\sigma}{\varepsilon} \quad,\left(\sigma=\mathrm{E}^{*} \varepsilon\right)$
TensileStrain

$$
\begin{gathered}
\varepsilon=\sigma / \mathrm{E} \quad, \quad \varepsilon=\mathrm{P} / \mathrm{AE}, \quad \varepsilon=\delta \mathrm{L} / \mathrm{L},, \quad \delta \mathrm{l}=\varepsilon * \mathrm{~L}, \quad \delta \mathrm{~L}=\mathrm{PL} / \mathrm{AE} \\
\mathrm{G}=\frac{\text { Shear Stress }}{} \text { or }=\frac{\tau}{\phi}
\end{gathered}
$$

Poisson＇s Ratio：the ratio of lateral strain to the longitudinal strain is a constant for a given material，when the material is stressed within the elastic limit．This ratio is called Poisson＇s ratio and it is generally denoted by $(\mu)$ ．Hence mathematically，

$$
\text { Poisson's ratio, } \mu=\frac{\text { Londaturalnstrainain }^{2}}{}
$$

Two－dimensional figure ABCD ，subjected to two mutually perpendicular stress $\sigma_{1}$ and $\sigma_{2}$ ．

Let $\quad \sigma_{1}=$ normal stress in x－direction $\sigma_{2}=$ normal stress in y－direction

Consider the strain produced by $\sigma_{1}$


The stress $\sigma_{1}$ will produce strain in the direction of $x$ and also in the direction of $y$ ． the strain in the direction of x will be longitudinal strain and will be equal to

$$
\text { Longitudinal Strain }=\frac{\sigma 1}{}
$$

whereas the strain in the direction of $y$ will be laterat strain and will be equal to

$$
\text { Lateral Strain }=-\mu * \text { longitudinal strain }=\mu * \frac{\sigma 1}{E}
$$

Consider the strain produced by $\sigma_{2}$
The stress $\sigma_{2}$ will produce strain in the direction of $y$ and also in the direction of $x$ ． the strain in the direction of y will be longitudinal strain and will be equal to

$$
\text { Longitudinal Strain }=\frac{\sigma 2}{}
$$

whereas the strain in the direction of y will be lateral ${ }^{E}$ strain and will be equal to

$$
\begin{gathered}
\text { Lateral Strain }=-\mu * \text { longitudinal strain }=-\mu * \frac{\sigma 2}{\mathrm{E}} \\
\varepsilon_{⿱ ㇒ 日 勺} \equiv \text { Total strain in }=\text { direction }
\end{gathered}
$$

$\sigma 1 \quad \sigma 2$
Now total strain in the direction of $x$ due to stresses $\sigma_{1}$ and $\sigma_{2}, \quad \varepsilon_{1}=--\mu *-$ Similarity，

E E
total strain in the direction of y due to stresses $\sigma_{1}$ and $\sigma_{2}$ ，

$$
\varepsilon_{2}=--\mu *-
$$

## For three-Dimensional Stress System

The figure shows that a three-dimensional body subjected to three orthogonal normal stresses $\sigma_{1}, \sigma_{2}, \sigma_{3}$ acting in the directions of $x, y$ and $z$ respectively. Consider the strains produced by each stress separately.


The stress $\sigma_{1}$ will produce strain in the direction of $x$ and also in the directions of $y$ and z .

The strain in the direction of x will be $\sigma 1$ $\sigma 1$ , the strains in the direction y and z will E
be $\left(-\mu^{*}\right.$ _) Similarity,

E
$\sigma 2$ $\sigma 2$

The stress $\sigma_{2}$ will produce strain the direction of x and z each. The stress $\sigma_{3}$ will produce strain $\frac{-}{E}$ in the direction of $y$ and the strain of $\left(-\mu^{*}-\right)$ in the direction of $x$ and $y$ each. E
total strain in the direction of $x$ due to stresses $\sigma_{1}, \sigma_{2}$ and $\sigma_{3}$,

$$
\varepsilon_{1}={ }^{\sigma 1}-\mu *-\frac{\sigma 2}{-\mu *-}
$$

total strain in the direction of y due to stresses $\frac{\mathrm{E}}{\mathrm{E}} \sigma_{1}, \sigma_{2}^{E}$ and $\sigma_{3}$,

$$
\varepsilon_{2}={ }_{\text {E. }}^{\sigma 2}-\mu \frac{\sigma 3}{E}-\mu * \frac{\sigma 1}{E}
$$

total strain in the direction of $z$ due to stresses $\sigma_{1}, \sigma_{2}$ and $\sigma_{3}$,

$$
\varepsilon_{3}={ }_{\text {E_ }}^{\sigma 3 *-\mu *-} \begin{gathered}
\sigma 1 \\
E
\end{gathered}
$$

Modulus of rigidity $G$ is related to the modulus of elasticity $E$ and Poisson's ratio $\mu$ by

$$
\mathrm{G}=\frac{\mathrm{E}}{2(1+w)}
$$

Problem 1. A rod 150 cm long and of diameter 2.0 cm is subjected to an axial pull of 20 kN . If the modulus of elasticity of the material of the $\operatorname{rod}$ is $2 * 10^{5} \mathrm{~N} / \mathrm{mm}^{2}$. Determine:

1- The stress
2- The strain
3- The elongation of the rod.

## Solution

Given: length of the rod, $\mathrm{L}=150 \mathrm{~cm}$
Dimeter of the rod $=2.0 \mathrm{~cm}=20 \mathrm{~mm}$

Area $=3.14 / 4\left(20^{2}\right)=314.16 \mathrm{~mm}^{2}$
Axial pull force $\quad \mathrm{P}=20 \mathrm{kN}=20000 \mathrm{~N}$
Modulus of elasticity $\mathrm{E}=2 * 10^{5} \mathrm{~N} / \mathrm{mm}^{2}$

1- $\quad$ Stress, $\sigma=\mathrm{P} / \mathrm{A}=20000 / 314.16=63.662 \mathrm{~N} / \mathrm{mm}^{2}$
Ans
2- $\quad$ Strain, $E=\sigma / \varepsilon$

$$
\varepsilon=\sigma / \mathrm{E}=63.662 /\left(2^{*} 10^{5}\right)=0.000318
$$

Ans

3- Elongation, $\quad \varepsilon=\mathrm{dL} / \mathrm{L}$

$$
\mathrm{dL}=\varepsilon * \mathrm{~L}
$$

$$
\mathrm{dL}=0.000318 * 150=0.0477 \mathrm{~cm}
$$

Problem 2. A square steel rod $20 \mathrm{~mm} * 20 \mathrm{~mm}$ in section is to carry an axial load of 100 kN . Calculate the shortening in a length of 50 mm . assume $\mathrm{E}=2.14^{*} 10^{8}$ $\mathrm{kN} / \mathrm{m}^{2}$.

## Solution:

$$
\begin{aligned}
& \text { Area }=0.02 * 0.02=0.0004 \mathrm{~m}^{2}=4 * 10^{-4} \mathrm{~m}^{2} \\
& \text { Length }=\mathrm{L}=50 \mathrm{~mm}=0.05 \mathrm{~m} \\
& \text { Load }=\mathrm{P}=100 \mathrm{kN} \\
& \mathrm{E}=2.14 * 10^{8} \mathrm{kN} / \mathrm{m}^{2} \\
& \text { Shortening of the rod } \quad \delta \mathrm{L}=? \\
& \sigma_{\mathrm{c}}=\mathrm{P} / \mathrm{A}=100 / 0.0004=250000 \mathrm{kN} / \mathrm{m}^{2} \\
& \mathrm{E}=\sigma / \varepsilon \quad=\varepsilon=\sigma / \mathrm{E}=250000 / 2.14 * 10^{8}=\delta \mathrm{L} / \mathrm{L} \\
& \delta \mathrm{~L}=\left(250000 / 2.14 * 10^{8}\right) * \mathrm{~L} \\
& \delta \mathrm{~L}=\left(250000 / 2.14 * 10^{8}\right) * 0.05 \\
& \delta \mathrm{~L}=0.0000584 \mathrm{~m} \quad \text { or } 0.0584 \mathrm{~mm}
\end{aligned}
$$

Hence, the shortening of the rod $=0.0584 \mathrm{~mm}$.
Ans.

Problem 3. The following observations were made during a tensile test on mild steel specimen 40 mm in diameter and 200 mm long. Elongation with 40 kN load (within limit of proportionality). $\delta \mathrm{L}=0.0304 \mathrm{~mm}$, Yield load $=161 \mathrm{kN}$, Maximum load $=$ 242 kN . Length of specimen at fracture $=249 \mathrm{~mm}$

## Determine the followings:

1- Young's modulus of elasticity
2- Yield point stress
3- Ultimate stress
4- Percentage elongation.

## Solution:

1- $\mathrm{E}=$ ?

$$
\begin{aligned}
& \sigma=\mathrm{P} / \mathrm{A}=40 /(3.14 / 4)\left(0.04^{2}\right)=3.18 * 10^{4} \mathrm{kN} / \mathrm{m}^{2} \\
& \varepsilon=\delta \mathrm{L} / \mathrm{L}=0.0304 / 200=0.000152 \\
& \mathrm{E}=\sigma / \varepsilon=3.18 * 10^{4} / 0.000152=2.09 * 10^{8} \mathrm{kN} / \mathrm{m}^{2}
\end{aligned}
$$

Ans.

2- Yield point stress = yield point load / Area

$$
=161 /(3.14 / 4) *\left(0.04^{2}\right)=12.8 * 10^{4} \mathrm{kN} / \mathrm{m}^{2}
$$

Ans.

3- Ultimate stress $=$ Max. Load $/$ Area $=242 /(3.14 / 4) *\left(0.04^{2}\right)$

$$
=19.2 * 10^{4} \mathrm{kN} / \mathrm{m}^{2}
$$

Ans.

4- Percentage elongation $=$ Strain $=\delta \mathrm{L} / \mathrm{L}=(\mathrm{Lf}-\mathrm{Lo}) / \mathrm{Lo}$

$$
=(249-200) / 200
$$

$$
=0.245 * 100=24.5 \%
$$

