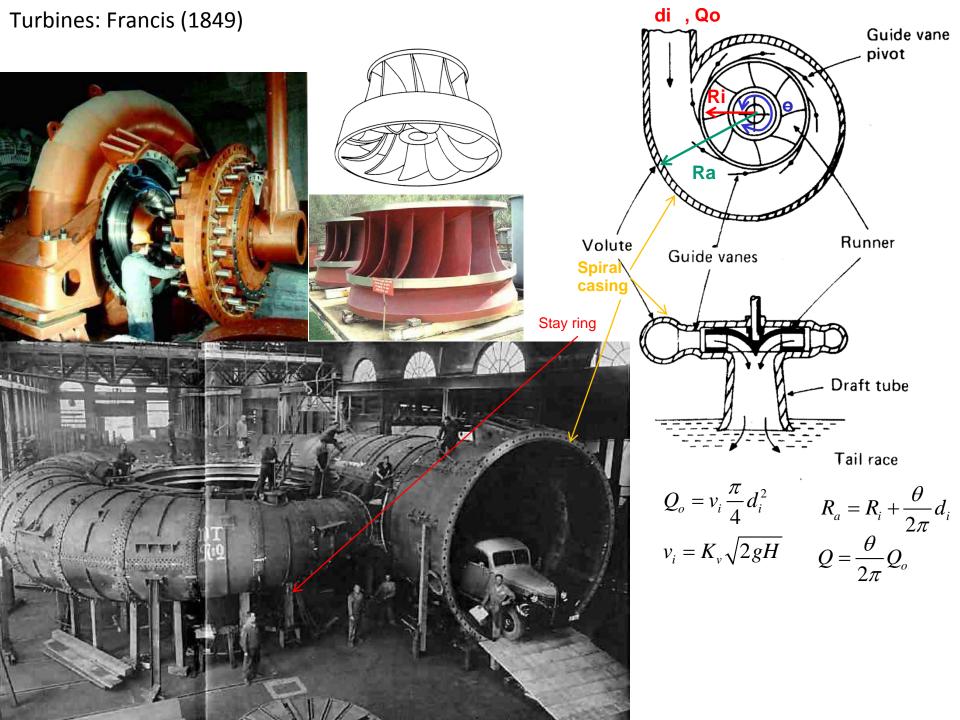
Lecture on Francis Turbine

by

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Draft Tube:

Static pressure ($P/\gamma+Z$) gradually decreases when water glides over the runner blades. Water coming out of the runner posses large amount of Kinetic energy and pressure at runner outlet, which is less than atmospheric pressure.

- (1) It makes possible to install the turbine above tail race without loss of head
- (2) The pressure at the exit of the draft tube is atmospheric.
- (3) Avoid cavitation, arrest separation of water.

Apply Bernouli's theory at runner exit (2) and exit of draft tube (4)

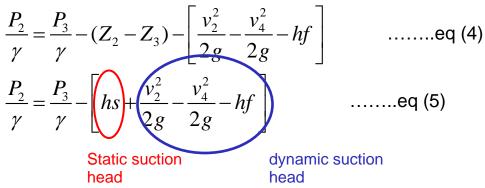
$$\frac{P_2}{\gamma} + \frac{v_2^2}{2g} + Z_2 = \frac{P_4}{\gamma} + \frac{v_4^2}{2g} + Z_4 + hf \quad \text{hf = head loss} \quad \dots \text{eq (1)}$$

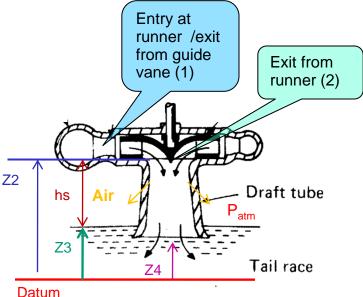
Apply Bernouli's theory at free surface (3) and exit of draft tube (4)

$$\frac{P_3}{\gamma} + \frac{v_3^2}{2g} + Z_3 = \frac{P_4}{\gamma} + \frac{v_4^2}{2g} + Z_4 \qquad \begin{array}{c} \text{Same, as} \\ \text{difference} \\ \text{is negligible} \end{array} \qquad \dots \text{eq (2)}$$

$$\frac{P_4}{\gamma} = \frac{P_3}{\gamma} + (Z_3 - Z_4) \qquad \dots \text{eq (3)}$$

Replacing the above value to eq (1)



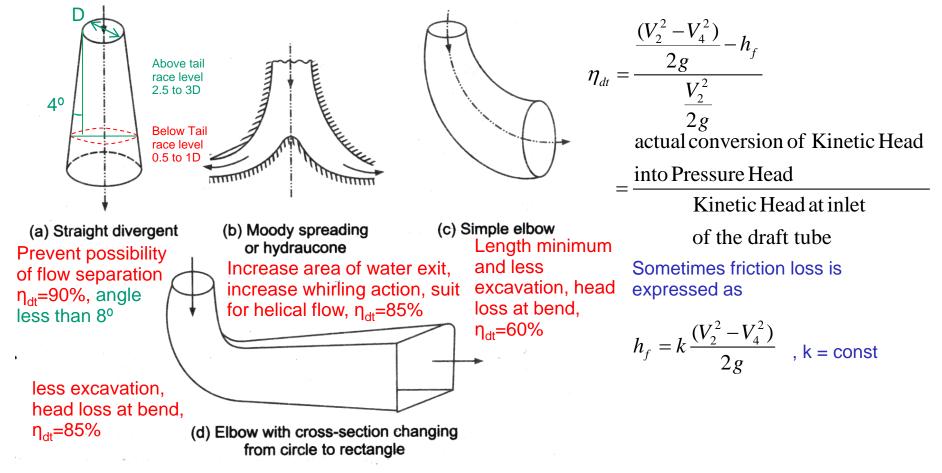


Draft Tube:

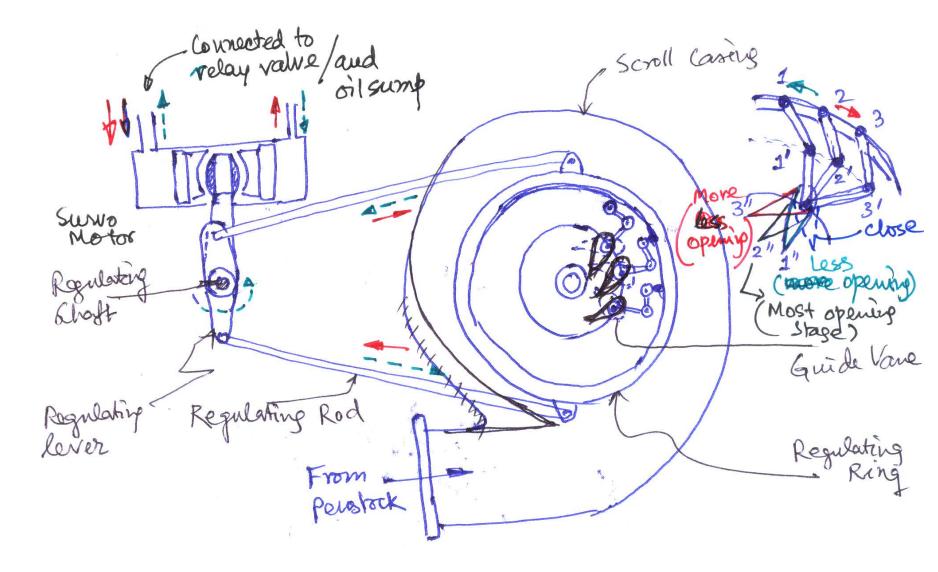
(4) Static pressure (P/ γ +Z) at runner outlet (at the level of 2) is less than atmospheric pressure by an amount equal to the static and dynamic suction head. (5) Velocity of water at outlet of runner is very high (3-15% of net working head, for high specific speed it is 45% in case of kaplan turbine), by employing draft tube recovers this wasted KE is utilized which increases efficiency of the turbine.

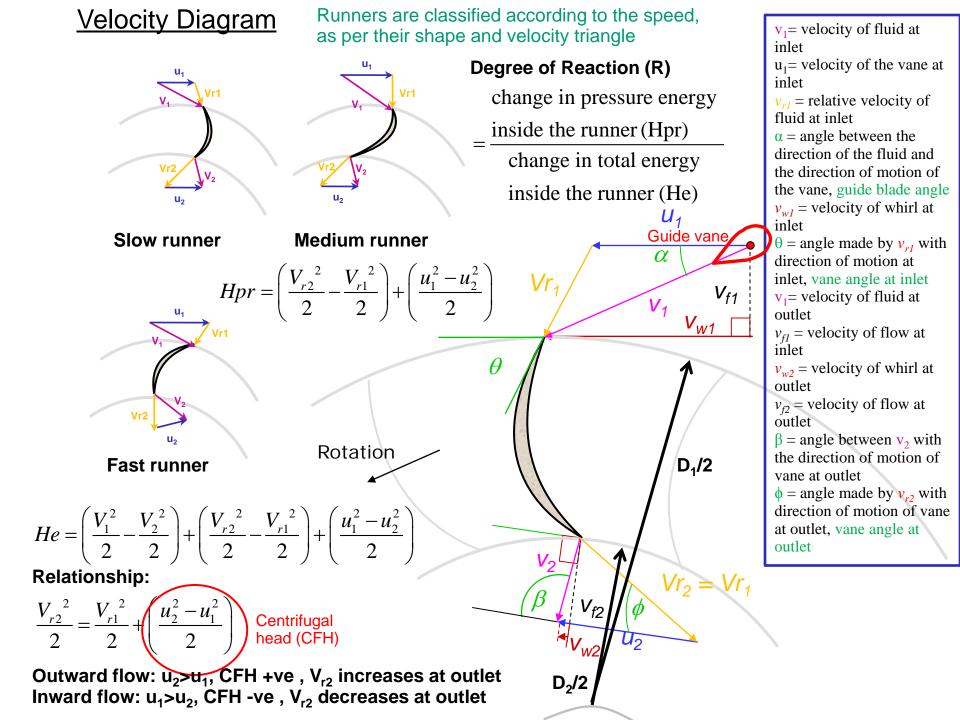
(6) Prevent splashes of water coming out of the runner.

Efficiency of the draft tube is expressed as



Governing Mechanism





Francis Turbine Equations

Working Proportions:

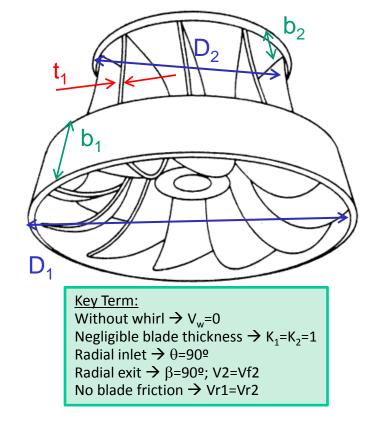
 (1) Ratio to width to diameter (n')= b1/D1 = 0.1 to 0.45, for slow runner flow is predominantly radial and exit is axial

(2) Speed ratio (Ku) =
$$u_1 / \sqrt{2gH}$$
 = 0.6 to 0.9

(3) Flow ratio (kf) = $V_{f1} / \sqrt{2gH} = 0.15$ to 0.30

(4) Coeff of velocity (kv)=
$$V_1 / \sqrt{2gH} = 0.97$$
 to 0.99

- Design Parameters: Head (H), running speed (N), Power output (P) is required size of the runner and its vane angle is to be find out.
- (1) Assume probable value of $\eta_h,\,\eta_o,\,n'$ and ratios (Ku, kf, kv)
- (2) Workout the mass or volumetric flow rate, where shaft power P = $\eta_o \gamma QH$



(3) Flow opening area $A_1 = \pi D_1 b_1 - Zt_1 b_1$, $= \pi D_1 b_1 K_1$ where Z is number of blades, t_1 is thickness of runner at inlet, b_1 = width of the runner at inlet, and K_1 is vane thickness coeff = approx 0.95

(4) Flow velocity $V_{f1} = Q/\pi D_1 b_1 K_1$, by continuity equation $Q = \pi D_1 b_1 K_1 V_{f1} = \pi D_2 b_2 K_2 V_{f2}$ (5) $u1 = \pi DN/60$

- (6) Work done by turbomachine $= \rho Q(Vw_1u_1 \pm Vw_2u_2)$,
- (7) Hydraulic Efficiency (η_h) assuming radial exit = $V_{w1}u_1/gH$ [V_{w2} is zero]
- (8) Head supplied to turbine = Work Done or head utilized at runner + Kinetic head at exit

 $H_{t} = \frac{\rho Q (V w_{1} u_{1} \pm V w_{2} u_{2})}{\rho g Q} + \frac{V_{2}^{2}}{2g} \quad \text{again gross head} \quad H_{g} = H_{t} + \sum h_{l} \text{ whereas for pump} \quad H_{p} = H_{g} - \sum h_{l}$